In the Abstract

Please replace the Abstract as presented in the underlying International Patent Application No. PCT/DE2004/002192 with the following amended Abstract:

ABSTRACT

~~The invention relates to a~~ A turbine engine and a rotor for a turbine engine are provided. Several adjacent ~~impeller~~ turbine blades are positioned in the circumferential direction of the rotor, each ~~impeller~~ turbine blade having a blade root, each ~~impeller~~ turbine blade being fixed in a retainer groove, running in the circumferential direction of the rotor, by means of the blade root. Each ~~impeller~~ turbine blade may be introduced, by the blade roots thereof, into the retaining groove, by means of ~~an introduction~~ a filling groove, whereby the width of the ~~introduction~~ filling groove is matched is matched to the width of the blade roots. ~~According to the invention, the~~ The width of the blade roots (13, 14) and the width of the, or each ~~introduction~~ filling groove (17) in the circumferential direction is greater than half the width of a desired nominal blade pitch (18), whereby in the region of the, or each ~~introduction~~ filling groove (17), a first number of ~~impeller~~ turbine blades (11) with desired nominal blade pitch (18), is exchanged for a second number of ~~impeller~~ turbine blades (12), with increased blade pitch (19), the first number being greater than the second number.